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DEVELOPMENT OF DIGITAL COMPASS BASED ON AVR MICROCONTROLLER AND MEMS ACCELEROMETER-MAGNETOMETER MODULE LSM303DLHC

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In the paper, the digital compass based on AVR microcontroller has been developed using MEMS accelerometer-magnetometer LSM303DLHC. The structure and hardware of the digital compass have been developed. The calibration algorithms of the accelerometer and magnetometer of the LSM303DLHC module have been described. The calculation algorithm of the pitch and roll angles has been described. The heading calculation algorithm of the digital compass without tilt compensation and with tilt compensation has been described. The operation algorithm and embedded software of the digital compass have been developed. The model and prototype of the digital compass have been created. The simulation and operation test of the developed digital compass have been performed.

Key words: digital compass, Arduino Uno R3, AVR microcontroller ATmega328P-PU, accelerometer, magnetometer, MEMS module LSM303DLHC, Proteus Design Suite, C programming language, Arduino IDE, embedded software.

Introduction

The development of electronics and micro-electro-mechanical systems technology has led to the appearance of integral motion sensors, such as accelerometers, gyroscopes, magnetometers etc. Magnetometer is a device for measuring the magnetic field strength (or intensity of one and more magnetic field components). MEMS magnetometers are widely used in handheld electronic devices, such as: smartphones, tablets, cars, robotics, etc. They are mostly a part of complex navigation systems, and in combination with an accelerometer and/or gyroscope represent an inertial navigation system that can accurately determine a location in three-dimensional space. Nowadays, the market has a wide choice of two- and three-axis integrated electronic compasses and modules with digital interfaces that contain a three-axis accelerometer and three-axis magnetometer [1,2]. The aim of the work is to develop the tilt-compensated digital compass based on AVR microcontroller and using MEMS module (accelerometer-magnetometer) LSM303DLHC and 128×64 graphic LCD ST7920.

Determination of the Earth's magnetic field strength and compass operation principle

For a more complete understanding of the compass operation principle, consider the basics of the magnetism theory and the principles for determination of the Earth's magnetic field vector direction (heading or azimuth).

The magnetic field strength of the Earth is from 0.5 to 0.6 Gauss. The Earth's magnetic field vector is parallel to its surface. It always orientates to the north magnetic pole. In the northern hemisphere, this field is directed downwards. At the equator, it is directed horizontally and in the southern hemisphere it is directed upwards [3, 4].



Fig. 1. Earth's magnetic field

For the creation of the tilt-compensated electronic compass, a 3-axis magnetic sensor (magnetometer) and a 3-axis accelerometer sensor (accelerometer) are needed. The accelerometer is used to measure the pitch and roll angles for the tilt compensation.

The magnetic sensor is used to measure the Earth's magnetic field strength and determine the heading angle relative to the magnetic north pole. [5, 6].

Terminology

In the compass the aviation terminology is used to determine the coordinates of the device and the three angles: pitch, roll and heading as shown in Fig. 2.



Fig. 2. Body coordinates and attitude angles

In Fig. 2, coordinates of the device X_b , Y_b and Z_b are directed forward, right, down based on the rule of the right hand. Three attitude angles (pitch, roll and heading) are referenced to the local horizontal plane that is perpendicular to the earth's gravity.

• Heading or azimuth (φ) is the angle between the Xb-axis and the north magnetic pole on the horizontal plane. It is measured in a clockwise direction from the top view of the device.

• Pitch (ρ) is the angle between the X_b -axis and the horizontal plane. When the device is rotated around the Y_b -axis, the axis moves upward, is positive and increases.

• Roll (γ) is the angle between the Y_b -axis and the horizontal plane. When the device is rotated around the X_b -axis, the Y_b -axis moves downwards. Roll is positive and increases.

Compass heading calculation

When the device is at an aligned position, pitch and roll angles are equal to 0° . Then, the angle can be determined directly as shown in Fig. 3.



Fig. 3. Heading (azimuth) calculation

The Earth's magnetic field (H) has a horizontal component H_h which points to the north magnetic pole of the Earth. In practice, for the determination of the heading (vector direction) of the Earth's magnetic field (H) the strength of its two components X_h and Y_h must be measured and angle between them has to be calculated (Fig. 3).

These components can be measured by the magnetometer. Then, the heading angle (φ) is calculated by the following formula:

$$f = \arctan \frac{\partial Y_h}{\partial X_h} \frac{\partial}{\partial x_h}$$
(1)

In Fig. 2, when the device body X_b -axis is parallel to H_h that indicates the north magnetic pole, then X_h is max and Y_h is 0, so the heading (φ) equals to 0°. The clockwise rotation of the device on the horizontal plane leads to the increase of the heading. When $X_h = 0$ and $Y_h = \min$, then heading angle $\varphi = 90^\circ$. Continuing the rotation to $X_h = \min$ and $Y_h = 0$, then heading = 180°. After a 360° rotation, the user will see a centered circle if the values of X_h and Y_h are obtained from the measurements of the magnetic sensor.

Tilt compensation

If the handheld device is tilted, the angles of the pitch and roll are not equal to 0° , as shown in Fig. 4. The pitch and roll can be measured by a three-axis accelerometer. Therefore, the values measured by the magnetic sensor X_M , Y_M , and Z_M must be offset to get X_h and Y_h , as shown in the eq. 2. And then the eq. 1 is applied to calculate the heading.



Fig. 4. Handheld device at tilted position

$$X_{h} = X_{M} \cos(\mathbf{r}) + Z_{M} \sin(\mathbf{r})$$
⁽²⁾

 $Y_{h} = X_{M} \sin(\mathbf{g})\sin(\mathbf{r}) + Y_{M} \cos(\mathbf{g}) - Z_{M} \sin(\mathbf{g})\cos(\mathbf{r}), \qquad (3)$

where X_M , Y_M and Z_M – values measured by the magnetometer.

LSM303DLHC accelerometer calibration algorithm

All STMicroelectronics MEMS accelerometers are calibrated that is sufficient for the most applications. For getting the heading accuracy less than 2°, the following calibration procedure is required. After the module installation into a handheld device, it must be re-calibrated. The accelerometer is recalibrated for determination of the bias value and scale factor, and the error matrix relative to the device axes X_b , Y_b and Z_b . The ratio between the normalized values A_{xl} , A_{yl} , and A_{zl} and the measured accelerometer values A_x , A_y , and A_z is determined as:

where $[A_m] - 3 \times 3$ misalignment matrix between the accelerometer sense axes and the device axes; A_SCi (i = x, y, z) is the scale factor and A_OSi is the offset.

The accelerometer calibration is needed for determination of the 12 parameters from ACC_{10} to ACC_{33} . So that with any given raw data arbitrary positions, normalized values can be obtained [7, 8].

Calibration can be done at the 6 stationary positions. The accelerometer data in each position known as A_{xl} , A_{yl} , and A_{zl} should be collected during 5–10 sec. Then, the method of least squares can be used to get the 12 optimal calibration parameters for the accelerometer.

LSM303DLHC magnetometer calibration algorithm

The LSM303DLHC magnetometer has the measurement resolution of the magnetic field 8 mG (milligauss) at VDD = +3V. The average magnitude of the horizontal component of the magnetic field is approximately in the range of 200 mG (more on the equator, less closer to the magnetic poles). Therefore, the expected compass heading accuracy is approximately 2.3° [= arctg (8/200)]. Within the range ±1.3 gauss, the LSM303DLHC magnetic sensor sensitivity is 1055 LSB/gauss for the X/Y axes and 950 LSB/gauss for the Z-axis.

The relationship between the normalized M_{xI} , M_{yI} , and M_{zI} and the measured values of the magnetometer can be expressed by the following equation:

Equation 4:

$$\begin{split} & \stackrel{e}{\Theta}_{x1} \stackrel{u}{\mathbf{u}}_{e} = \begin{bmatrix} M_{-}m \end{bmatrix}_{3'3} \stackrel{e}{\mathbf{e}} & 0 & 1/M_{-}SC_{x} & 0 & 0 & \dot{\mathbf{u}} \\ & \stackrel{e}{\Theta}_{y1} \stackrel{u}{\mathbf{u}} = \begin{bmatrix} M_{-}m \end{bmatrix}_{3'3} \stackrel{e}{\mathbf{e}} & 0 & 1/M_{-}SC_{y} & 0 & \dot{\mathbf{u}} \\ & \stackrel{e}{\mathbf{e}} M_{z1} \stackrel{d}{\mathbf{g}} & \stackrel{e}{\mathbf{e}} & 0 & 0 & 1/M_{-}SC_{z} \stackrel{d}{\mathbf{g}} \\ & \stackrel{e}{\mathbf{e}} M_{x} - M_{-}OS_{x} \stackrel{u}{\mathbf{u}}_{x} \stackrel{e}{\mathbf{e}} MR_{11} & MR_{12} & MR_{13} \stackrel{u}{\mathbf{u}}_{x} \stackrel{e}{\mathbf{e}} M_{x} - MR_{10} \stackrel{u}{\mathbf{u}} \\ & \stackrel{e}{\mathbf{e}} M_{x} - M_{-}OS_{y} \stackrel{u}{\mathbf{u}}_{y} \stackrel{e}{\mathbf{e}} \stackrel{e}{\mathbf{e}} MR_{21} & MR_{22} & MR_{23} \stackrel{u}{\mathbf{u}}_{y} \stackrel{e}{\mathbf{e}} M_{y} - MR_{20} \stackrel{u}{\mathbf{u}}, \\ & \stackrel{e}{\mathbf{e}} M_{z} - M_{-}OS_{z} \stackrel{u}{\mathbf{g}}_{y} \stackrel{e}{\mathbf{e}} MR_{31} & MR_{32} & MR_{33} \stackrel{u}{\mathbf{g}}_{y} \stackrel{e}{\mathbf{e}} M_{z} - MR_{30} \stackrel{u}{\mathbf{g}} \end{split}$$
(5)

where $[M_m]$ is a 3×3 misalignment matrix between the magnetic sensor axes and the device axes; M_SCi (i = x, y, z) is the scale factor and M_OSi is the offset caused by hard-iron distortion; $[M_si] - 3\times3$ matrix caused by soft-iron distortion.

The magnetometer calibration is used for determination of the parameters from MR_{10} to MR_{33} , thus with any known measured values at arbitrary positions. Thus, the normalized values can be calculated for any raw measurements at arbitrary positions [7, 8].

Pitch and roll calculation algorithm

Let's suppose that the LSM303DLHC chip is installed on the device, as shown in Fig. 5.



of the electronic compass

 X_b , Y_b and Z_b are the device body axes which are directed forward, right and down, respectively. The accelerometer and magnetometer sense axes are $X_{A,M}$, $Y_{A,M}$, and $Z_{A,M}$, respectively. The sign of $Y_{A,M}$ and $Z_{A,M}$ from the sensor measurements has to be reversed to make the same direction of the sense axes as the axes of the device. Pitch/roll/heading angles are referenced to the local horizontal plane which is perpendicular to the Earth's gravity.

When the device is at the 3D arbitrary position X'_b , Y'_b and Z'_b , there are several procedures for the device rotation from the local coordinate system X_b . Y_b and Z_b , as shown in Fig. 8. The different rotation procedures lead to the different rotation matrix.

Firstly, the device has to be rotated around the Z_b -axis clockwise at the angle (φ) with a view from the origin to downwards. Then, turn the device around the Y_b -axis at the angle (φ), the X_b -axis moves to the top. Then, the device is rotated around X_b to the angle (γ), Y_b moves to the bottom. The new coordinates of the device will be X'_b , Y'_b and Z'_b , as shown in Fig. 5.



Fig. 6. Rotation procedure

Then, each matrix of rotation is:

=

$$\begin{aligned} \hat{\mathbf{e}} & \cos \mathbf{f} & \sin \mathbf{f} & 0 \mathbf{\hat{u}} \\ R_{\mathbf{y}} &= \hat{\mathbf{e}}^{\mathbf{\hat{e}}} \cdot \sin \mathbf{f} & \cos \mathbf{f} & 0 \mathbf{\hat{u}}^{\mathbf{\hat{u}}}, \\ \hat{\mathbf{e}} & 0 & 0 & 1 \mathbf{\hat{e}} \end{aligned} \tag{6}$$

$$R_{\rm r} = \begin{pmatrix} \dot{e}\cos r & 0 & -\sin r \dot{u} \\ \dot{e} & 0 & 1 & 0 & \dot{u}, \\ \dot{e}\sin r & 0 & \cos r & \dot{e} \end{pmatrix}$$
(7)

$$\begin{aligned} \hat{e}_{g} &= \hat{e}_{\hat{e}}^{0} \cos g \quad \sin g \stackrel{\acute{u}}{\underline{u}}. \\ \hat{e}_{0} &- \sin g \quad \cos g \stackrel{\acute{u}}{\underline{u}}. \end{aligned} \tag{8}$$

The relationship between $X'_{b}/Y'_{b}/Z'_{b}$ and $X_{b}/Y_{b}/Z_{b}$ is the following:

$$\begin{array}{cccc} & \stackrel{\acute{e}X_{b}}{\overset{i}{\upsilon}} \dot{u} & \stackrel{\acute{e}X_{b}}{\overset{i}{\upsilon}} \dot{u} \\ \stackrel{\acute{e}Y_{b}}{\overset{i}{\upsilon}} \overset{i}{\underline{u}} = R_{g}R_{r}R_{y} & \stackrel{\acute{e}Y_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} = \\ \stackrel{\acute{e}Z_{b}}{\overset{i}{\upsilon}} \overset{i}{\underline{u}} = R_{g}R_{r}R_{y} & \stackrel{\acute{e}X_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} = \\ \stackrel{\acute{e}Z_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} & \stackrel{\acute{e}Z_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} \\ \stackrel{\acute{e}Z_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} & \stackrel{\acute{e}Z_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} \\ \stackrel{\acute{e}C}{\overset{cosf}} sinr sing - cosg sinf & cosg cosf + sinr sing sinf & cosr sing \overset{i}{\underline{u}} \overset{\acute{e}X_{b}}{\overset{i}{\vartheta}} \overset{i}{\underline{u}} \\ \stackrel{\acute{e}C}{\overset{e}cosf} sinr cosg + sing sinf & - sing cosf + sinr cosg sinf & cosr cosg \not{e}I_{b} & \stackrel{i}{\underline{u}} \end{array}$$
 (9)

In the local horizontal plane (Fig. 5) $X_b = Y_b = 0$, $Z_b = +1g$. At $X'_b/Y'_b/Z'_b$, the accelerometer raw measurements are A_x , A_y and A_z . They are signed integer in terms of LSBs. Let A_{x1} , A_{y1} and A_{z1} be the normalized values after using the accelerometer calibration parameters for A_x , A_y , and A_z . So A_{x1} , A_{y1} and A_{z1} become the floating point values less than 1 in terms of g and the root of the sum of their squared values should be equal to 1 when the accelerometer is still. Then the equation 9 will be:

$$\begin{array}{lll} \acute{e}A_{x1}\grave{u}&\acute{e}&\cos r\cos y&\cos r\sin f&-\sin r&\check{u}\acute{e}0\grave{u}\\ \acute{e}A_{y1}\acute{u}&=\stackrel{\circ}{e}\cos f\sin r\sin g-\cos g\sin f&\cos g\cos f+\sin r\sin g\sin f&\cos r\sin f&\overset{\circ}{u}\overset{\circ}{e}0\overset{\circ}{u}\\ \acute{e}A_{z1}\acute{g}&\stackrel{\circ}{e}\cos f\sin r\cos g+\sin g\sin f&-\sin g\cos f+\sin r\cos g\sin f&\cos r\cos f&\stackrel{\circ}{e}\acute{l}\acute{g}\acute{l}&\acute{g} \end{aligned}$$

Thus, pitch and roll angles can be calculated by the following equations:

Pitch:
$$\mathbf{r} = \arcsin(-A_{x1}),$$
 (11)

Roll:
$$\mathbf{g} = \arcsin\left(A_{vl} / \cos \mathbf{r}\right)$$
. (12)

When pitch = $\pm 90^{\circ}$, roll has to be set to 0° for avoiding singularity. The function of arcsin has a good linearity from -45° to +45°, so the pitch and roll calculation accuracy degrades if tilt angles exceed this range.

The normalized value of the accelerometer A_{zl} is not used for the pitch and roll calculation, but it can be used to check the equality of the magnitude $|A| = \sqrt{A_{x1}^2 + A_{y1}^2 + A_{z1}^2}$ to 1. If it does not equal to 1, then there is the detected linear acceleration or angular acceleration.

Heading calculation algorithm

For the heading calculation, the values measured by the 3-axis magnetometer should be normalized using the calibration parameters of the magnetometer and reflected on the horizontal plane for tilt compensation.



Fig. 6. Heading calculation where: H – horizontal component of the Earth's magnetic field and Zb points into the page

If the device rotates from $X_b/Y_b/Z_b$ to $X''_b/Y''_b/Z''_b$ by roll angle rotation followed by pitch angle rotation, then:

$$\begin{aligned} & \stackrel{e}{e}X_{b}\stackrel{i}{u} & \stackrel{e}{e}X_{b}\stackrel{i}{u}\stackrel{i}{e} e \cos r & 0 & \sin r & \stackrel{i}{u}\stackrel{e}{e}X_{b}\stackrel{i}{u} \\ & \stackrel{e}{e}Y_{b}\stackrel{i}{u} = R_{g}^{-1}R_{r}^{-1}\stackrel{e}{e}Y_{b}\stackrel{i}{u}\stackrel{i}{u} = \stackrel{e}{e}\sin g\sin r & \cos g & -\sin g\cos r & \stackrel{i}{u}\stackrel{e}{e}Y_{b}\stackrel{i}{u}. \end{aligned}$$
(12)
$$& \stackrel{e}{g}Z_{b}\stackrel{i}{g}\stackrel{i}{e} & \stackrel{e}{e}Z_{b}\stackrel{i}{u}\stackrel{i}{g} = \cos g\sin r & \sin g & \cos g\cos r & \stackrel{i}{g}\stackrel{e}{e}Z_{b}^{i}\stackrel{i}{u} \end{aligned}$$

Applying the calibration parameters correction into magnetic sensor raw measurements M_x , M_y , and M_z at new positions $X''_b/Y''_b/Z''_b$, the normalized magnetic sensor measurements M_{x1} , M_{y1} and M_{z1} will be obtained.

The values of M_x , M_y and M_z are signed integers in the terms of LSBs. The floating point values M_{xl} , M_{yl} and M_{zl} are less than 1 in the terms of the magnetic field strength. The square root of the sum squared values has to be equal to 1 when there is no external interference magnetic field. Thus, using the eq. 12 the tilt compensated magnetic sensor measurements M_{x2} , M_{y2} , M_{z2} are obtained:

 M_x , M_y and M_z which are signed integer in terms of LSBs. M_{x1} , M_{y1} and M_{z1} are floating point values less than 1 in terms of the magnetic field strength. The square root of the sum squared values

should be equal to 1 when the there is no external interference magnetic field. Then from the eq. 12, tilt-compensated magnetic sensor measurements M_{x2} , M_{y2} , and M_{z2} can be obtained as

$$M_{x2} = M_{x1} \cos r + M_{z1} \sin r , \qquad (13)$$

$$M_{y2} = M_{x1} \sin g \sin r + M_{y1} \cos g - M_{z1} \sin g \cos r ,$$

$$M_{z2} = -M_{x1} \cos g \sin r + M_{y1} \sin g + M_{z1} \cos g \cos r .$$

Equation 14:

$$f = \arctan \frac{\partial M_{y2}}{\partial k_{x2}} \int_{M_{x2}}^{O} \dot{b} \text{ for } M_{x2} > 0 \text{ and } M_{y2}^{3} 0 =$$

$$= 180^{\circ} + \arctan \frac{\partial M_{y2}}{\partial k_{x2}} \int_{O}^{O} \dot{b} \text{ for } M_{x2} < 0 =$$

$$= 360^{\circ} + \arctan \frac{\partial M_{y2}}{\partial k_{x2}} \int_{O}^{O} \dot{b} \text{ for } M_{x2} > 0 \text{ i } M_{y2} \pounds 0 =$$

$$= 90^{\circ} \text{ for } M_{x2} = 0 \text{ and } M_{y2} < 0 \text{ or } = 270^{\circ} \text{ for } M_{x2} = 0 \text{ and } M_{y2} > 0.$$
(14)

The magnitude $|M| = \sqrt{M_{x2}^2 + M_{y2}^2 + M_{z2}^2}$ has to be also equal to 1. If not, it means that an external interference magnetic field is detected.

Development of digital compass

In Fig. 7, the block diagram of the digital compass is shown. The microcontroller reads the data from the 3-axis accelerometer for the pitch and roll angle calculation, and from the 3-axis magnetometer for the heading calculation.

The algorithm of the digital compass development consists of:

· Hardware design for accelerometer and magnetometer data acquisition;

• Accelerometer calibration for obtaining the parameters needed to convert the raw data from the accelerometer to normalized values for pitch and roll angle calculation;

• Magnetometer calibration for obtaining the parameters to convert the raw data from the magnetometer to the normalized values for the heading calculation;

Digital compass testing.



Fig. 7. Block diagram of the digital compass built on the accelerometer-magnetometer LSM303DLHC

LSM303DLHC is a sensor module manufactured by STMicroelectronics that combines 3-axis accelerometer and 3-axis magnetometer, Fig. 8.



Fig. 8. The board with LSM303DLHC module and voltage regulator

LSM303DLHC can be connected to the microcontroller through the I²C digital interface. In Fig. 9, the typical LSM303DLHC connection circuit to the microcontroller via the I²C interface is shown. The power supply for the LSM303DLHC operation is +3V and logic supply voltage can be in the range of 1.8 to 3.3 V.



Fig. 9. Typical connection circuit of the LSM303DLHC module to the microcontroller

The LSM303DLHC module is connected through the I^2C bus as a slave device. The microcontroller is a master device. In the digital compass design, the following has to be taken into account [9]:

Reserved pins have to be connected according to the LSM303DLHC datasheet.

• Ceramic power source decoupling capacitors should be placed as close to the VDD (pin 6) as possible.

- The selected microcontroller must have a built-in I^2C interface controller.
- · Power and logic supplies should be adjustable and with small noises.

Software of digital compass

Data acquisition from the accelerometer

If the pin 4 SA0_A is connected to voltage supply, the 7-bit I²C slave address of the accelerometer will be 0×19 . And otherwise if the pin 4 SA0_A is connected to ground, then the 7-bit I²C slave address of the accelerometer will be 0×18 .

After powering on the LSM303DLHC module, the two registers CTRL_REG1_A (20h) and CTRL_REG4_A (23h) must be configured. For the normal operation mode (ODR=50 Hz), the value 0×27 is written to the register CTRL_REG1_A. To setup the continuous mode of the data update, the value 0×40 has to be written to the CTRL_REG4_A register [10]. In the following block-diagram, the algorithm of the accelerometer data acquisition is shown, Fig. 10:



Fig. 10. The block-diagram of the data acquisition from the accelerometer

Data acquisition from the magnetometer LSM303DLHC

The magnetometer has the 7-bit I²C slave address $0\times1E$. After powering on the LSM303DLHC module, the two registers CRA_REG_M (00h) and MR_REG_M (02h) have to be configured. In order to do this, the value 0×14 has to be written to the CRA_REG_M register for changing the data output rate from 15 Hz to 30 Hz. The value 0×00 has be written to the register MR_REG_M to wake up the

magnetometer for the normal operation mode [10]. The range ± 1.3 gauss and data output rate (ODR) 30 Hz are sufficient for the digital compass.

The following block-diagram demonstrates the data acquisition algorithm from the magnetometer (Fig. 11):



Fig. 11 The block-diagram of the data acquisition from the LSM303DLHC magnetometer

The pointer of the register address of the magnetometer is automatically updated. After reading the register MR_REG_M (02h), the pointer will automatically increase by 1 to 03h that is the address of the register OUT_X_H_M. After reading the register OUT_X_H_M, the address pointer will increase by 1 to the address of the register OUT_X_L_M. The embedded software of the digital compass is compiled into a hex file that is flashed into the ATmega328 microcontroller of the Arduino Uno platform.



Fig. 12. Operation algorithm of the digital compass

Research Results and Their Analysis

The research results are graphically presented in Fig. 13–16. The designed circuit of the digital device is simulated in Proteus (Fig. 13) and tested on the developed prototype (Fig. 14).

In Fig. 14–15 the raw data obtained on the x-, y-, z-axes from the accelerometer Ax, Ay, Az and magnetometer Mx, My, Mz has been shown. The graphical view of the compass heading and direction – southeast (SE) is shown on the 128×64 graphic LCD ST7920 (Fig. 16).

Digital compass built on accelerometer+magnetomter LSM303DLHC



Fig. 13. Digital compass model developed in Proteus



Fig. 14. Accelerometer and magnetometer raw data displayed on the graphic LCD ST7920

💿 COM6 (Arduino/Genuino Mega or Mega 2560)					- 0	×
						Send
Mx: -290	My: 780 Mz:	147				^
Heading: 110	.395 degrees					
Tilt heading	g: 111.140 degree	8				
*** Accelero	ometer Raw Data *	r.k.				
Ax: -192	Ay: 1536	Az: -17024				
*** Magnetor	neter Raw Data **					
Mx: -291	My: 781 Mz:	147				
Heading: 110	.435 degrees					
Tilt heading	g: 111.236 degree					
*** Accelero	ometer Raw Data *	(k)				
Ax: -256	Ay: 1600	Az: -17216				
*** Magnetor	neter Raw Data **					
Mx: -289	My: 776 Mz:	146				
Heading: 110	.426 degrees					
Tilt heading	g: 111.166 degree	1				
*** Acceler						¥
Autoscroll [Show timestamp		Newline	🤟 9600 baud	~ Clea	r output

Fig. 15. Accelerometer and magnetometer raw data output on the serial monitor



Fig. 16. Graphical indication of the compass heading on the 128×64 *graphic LCD ST7920*

Conclusion

The digital compass based on the Arduino Uno platform using the MEMS module (accelerometermagnetometer) LSM303DLHC has been developed. The structure of the digital compass hardware has been developed. The algorithms for accelerometer and magnetometer calibration, for calculating roll and pitch angles have been developed. The algorithm of compass heading calculation without tilt compensation and with tilt compensation has been developed. The embedded software of the digital compass has been developed. The model and prototype of the digital compass have been created. The research and result analysis of the digital compass operation have been conducted.

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РОЗРОБКА ЦИФРОВОГО КОМПАСУ НА AVR МІКРОКОНТРОЛЕРІ ТА MEMS – МОДУЛІ АКСЕЛЕРОМЕТРА – МАГНІТОМЕТРА LSM303DLHC

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У роботі розроблено цифровий компас на MK AVR з використанням MEMC – модуля акселерометра – магнітометра LSM303DLHC. Розроблено структуру та спроектовано апаратне забезпечення цифрового компасу. Описано алгоритми калібрування акселерометра та магнітометра LSM303DLHC. Описано алгоритм обчислення кутів нахилу (кутів тангажу і крену) цифрового компасу. Описано алгоритм визначення курсу (напрямку) компаса без компенсації і з компенсацією нахилу. Розроблено алгоритм роботи та програмне забезпечення цифрового компасу. Створено модель та макет цифрового компасу. Проведено моделювання і тестування розробленого цифрового компасу.

Ключові слова: цифровий компас, апаратно-програмна платформа Arduino Uno R3, МК ATmega328P-PU, акселерометр, магнітометр, MEMC модуль LSM303DLHC, САПР Proteus Design Suite, мова програмування C, середовище програмування Arduino IDE для МК платформи Arduino, вбудоване програмне забезпечення.